

Aerodynamics and fluid-structure interaction simulation with AMROC

Part I

Ralf Deiterding

Aerodynamics and Flight Mechanics Research Group
University of Southampton
Highfield Campus
Southampton SO17 1BJ, UK
Email: r.deiterding@soton.ac.uk

Xiamen
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Outline

Fluid-structure coupling

- Approach

- Rigid body motion

- Thin elastic and deforming thin structures

- Real-world example

Train-tunnel aerodynamics

- Validation

- Passing trains in open space

- Passing trains in a double track tunnel

Summary

- Conclusions

Collaboration with

Finite volume methods

- ▶ Jose M. Garro Fernandez (University of Southampton)
- ▶ Stuart Laurence (Department of Aerospace Engineering, University of Maryland, College Park)
- ▶ Fehmi Cirak (Cambridge University)
- ▶ Sean Mauch, Joe Shepherd, Dan Meiron (California Institute of Technology)

Lattice Boltzmann methods

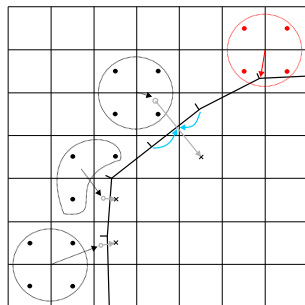
- ▶ Christos Gkoudesnes, Juan Antonio Reyes Barraza (University of Southampton)
- ▶ Stephen Wood (NASA)
- ▶ Kai Feldhusen, Claus Wagner (German Aerospace Center – DLR)
- ▶ Moritz Fragner (University of Applied Sciences Hannover, Germany)
- ▶ Cinar Laloglu (Marmara University, Turkey)

Construction of coupling data

- ▶ Moving boundary/interface is treated as a moving contact discontinuity and represented by level set [Fedkiw, 2002][Arienti et al., 2003]
- ▶ Efficient construction of level set from triangulated surface data with closest-point-transform (CPT) algorithm [Mauch, 2003]

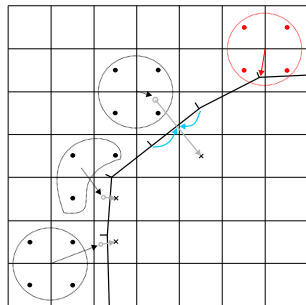
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Coupling conditions on interface

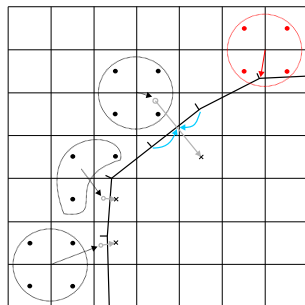
Viscous fluid:

$$\begin{aligned} u^S &= u^F \\ \sigma_{nm}^S &= \sigma_{nm}^F \end{aligned} \Big|_{\mathcal{I}}$$

$$\text{with } \sigma_{nm}^F = -p^F \delta_{nm} + \Sigma_{nm}^F$$

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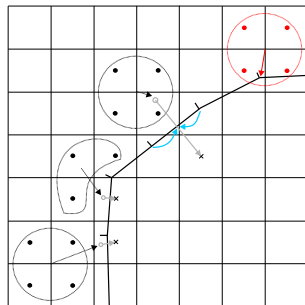
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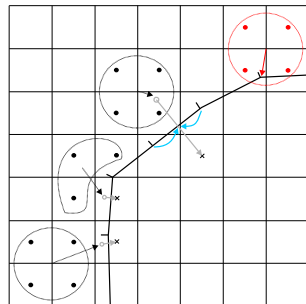
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[Deiterding and Wood, 2013]

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Closest point transform algorithm

The signed distance φ to a surface \mathcal{I} satisfies the eikonal equation [Sethian, 1999]

$$|\nabla\varphi| = 1 \quad \text{with} \quad \varphi|_{\mathcal{I}} = 0$$

Solution smooth but non-differentiable across characteristics.

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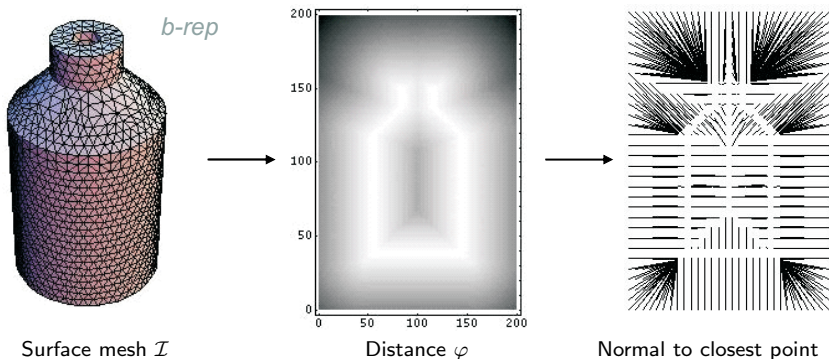
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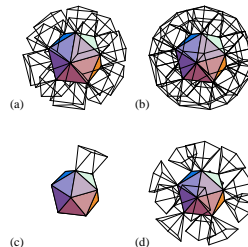
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The characteristic / scan conversion algorithm

1. Build the characteristic polyhedrons for the surface mesh

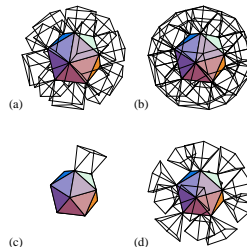
Characteristic polyhedra for faces, edges, and vertices



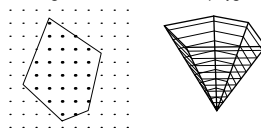
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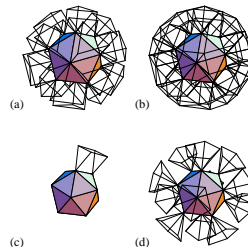
Slicing and scan conversion of apolygon



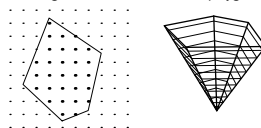
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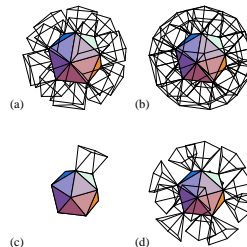
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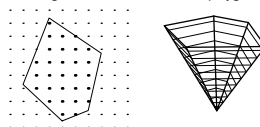
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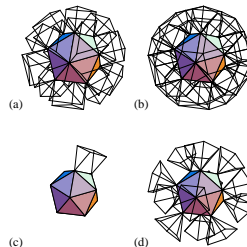
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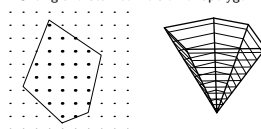
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4. Problem reduction by evaluation only within specified max. distance

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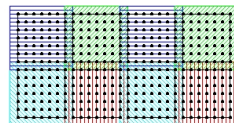
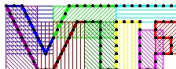
Slicing and scan conversion of apolygon



[Mauch, 2003], see also
[Deiterding et al., 2006]

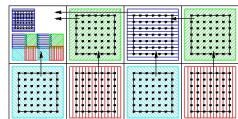
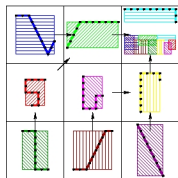
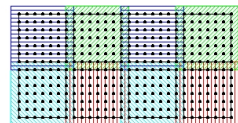
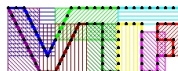
Eulerian/Lagrangian communication module

1. Put bounding boxes around each solid processors piece of the boundary and around each fluid processors grid



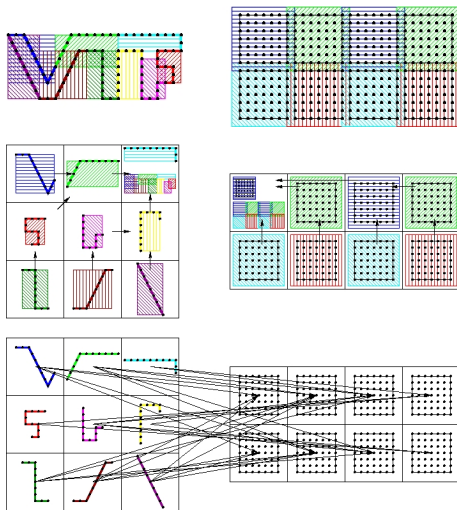
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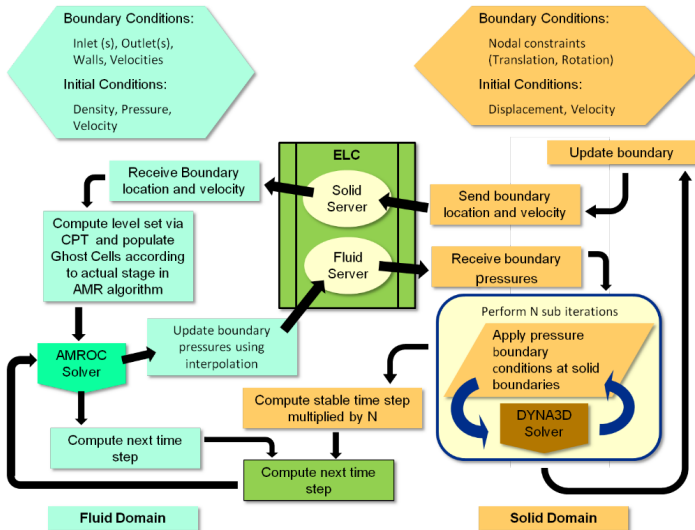


Eulerian/Lagrangian communication module

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2. Gather, exchange and broadcast of bounding box information
3. Optimal point-to-point communication pattern, non-blocking



Coupling elements



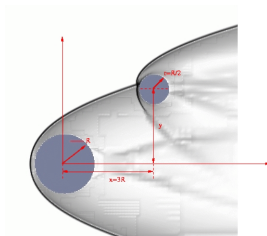
Proximal bodies in hypersonic flow

Flow modeled by Euler equations for a single polytropic gas with $p = (\gamma - 1) \rho e$

$$\partial_t \rho + \partial_{x_n}(\rho u_n) = 0, \quad \partial_t(\rho u_k) + \partial_{x_n}(\rho u_k u_n + \delta_{kn} p) = 0, \quad \partial_t(\rho E) + \partial_{x_n}(u_n(\rho E + p)) = 0$$

Numerical approximation with

- ▶ Finite volume flux-vector splitting scheme with MUSCL reconstruction, dimensional splitting



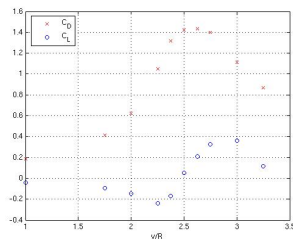
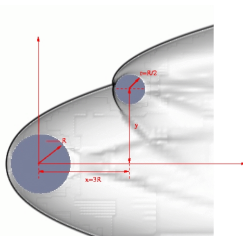
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Numerical approximation with

- ▶ Finite volume flux-vector splitting scheme with MUSCL reconstruction, dimensional splitting
- ▶ Spherical bodies, force computation with overlaid latitude-longitude mesh to obtain drag and lift coefficients $C_{D,L} = \frac{2F_{D,L}}{\rho v^2 \pi r^2}$
- ▶ inflow $M = 10$, C_D and C_L on secondary sphere, lateral position varied, no motion



Verification and validation

Static force measurements, $M = 10$:

[Laurence et al., 2007]

- Refinement study: $40 \times 40 \times 32$ base grid ,
up to without AMR up to $\sim 209.7 \cdot 10^6$
cells, largest run $\sim 35,000$ h CPU

I_{\max}	C_D	ΔC_D	C_L	ΔC_L
1	1.264		-0.176	
2	1.442	0.178	-0.019	0.157
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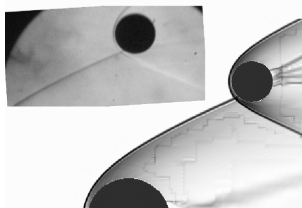
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- Comparison with experimental results: 3
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	Experimental	Computational
C_D	1.11 ± 0.08	1.01
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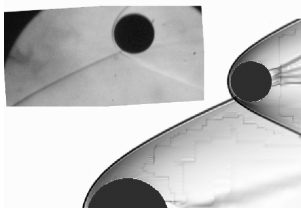
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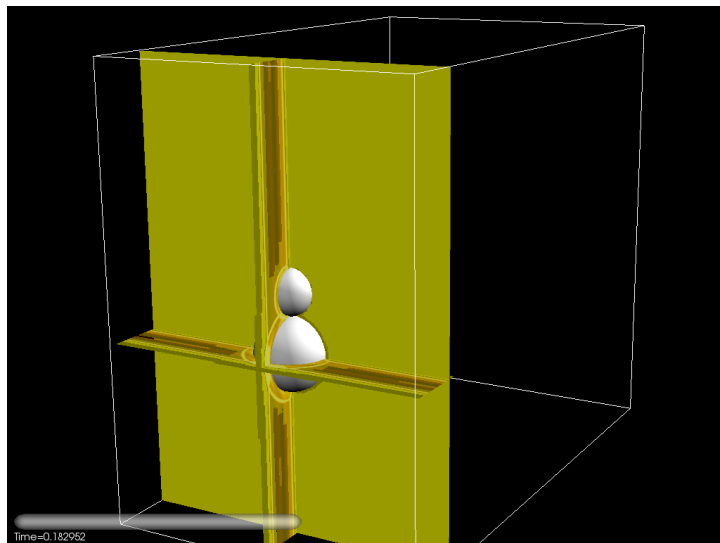
Dynamic motion, $M = 4$:

- Base grid $150 \times 125 \times 90$, two additional levels with $r_{1,2} = 2$
- 24,704 time steps, 36,808 h CPU on 256 cores IBM BG/P



[Laurence and Deiterding, 2011]

Schlieren graphics on refinement regions



Treatment of thin structures

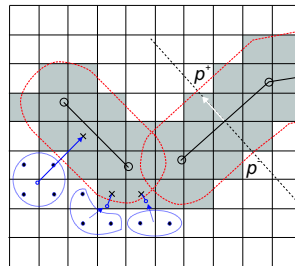
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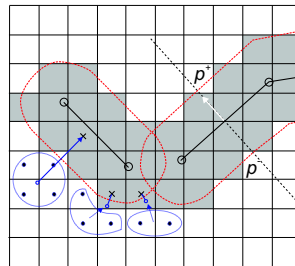
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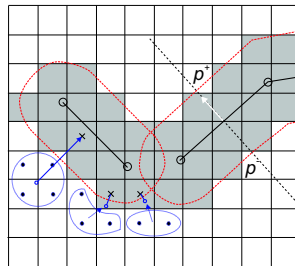
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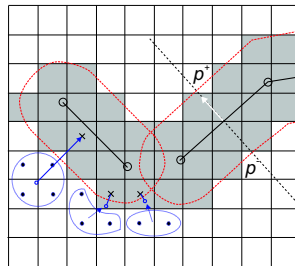
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- ▶ Use face normal in shell element to evaluate in $\Delta p = p^+ - p^-$
- ▶ Utilize finite difference solver using the beam equation



$$\rho_s h \frac{\partial^2 w}{\partial t^2} + EI \frac{\partial^4 w}{\partial \bar{x}^4} = p^F$$

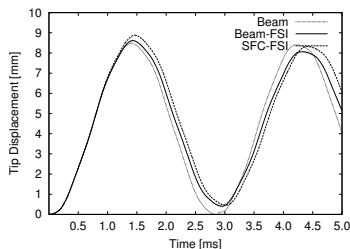
to verify FSI algorithms

FSI verification by elastic vibration

- ▶ Thin steel plate (thickness $h = 1$ mm, length 50 mm), clamped at lower end
- ▶ $\rho_s = 7600 \text{ kg/m}^3$, $E = 220 \text{ GPa}$, $I = h^3/12$, $\nu = 0.3$
- ▶ Modeled with beam solver (101 points) and thin-shell FEM solver (325 triangles) by F. Cirak

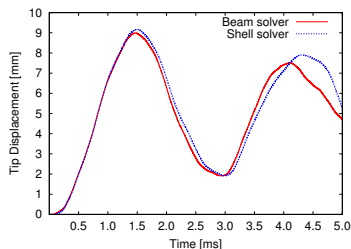
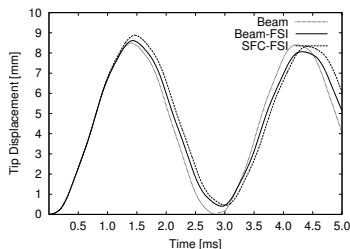
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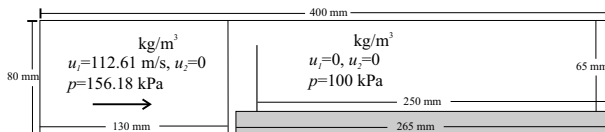
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- ▶ Left: Coupling verification with constant instantaneous loading by $\Delta p = 100 \text{ kPa}$
- ▶ Right: FSI verification with Mach 1.21 shockwave in air ($\gamma = 1.4$)



Shock-driven elastic panel motion

Test case suggested by [Giordano et al., 2005]

- ▶ Forward facing step geometry, fixed walls everywhere except at inflow

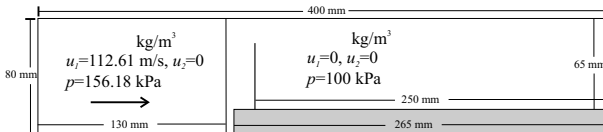


- ▶ SAMR base mesh $320 \times 64(\times 2)$, $r_{1,2} = 2$

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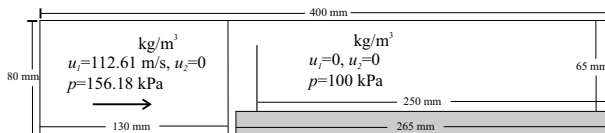


- ▶ SAMR base mesh $320 \times 64(\times 2)$, $r_{1,2} = 2$
- ▶ Intel 3.4GHz Xeon dual processors, GB Ethernet interconnect
 - ▶ Beam-FSI: 12.25 h CPU on 3 fluid CPU + 1 solid CPU
 - ▶ FEM-FSI: 322 h CPU on 14 fluid CPU + 2 solid CPU

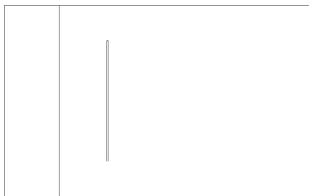
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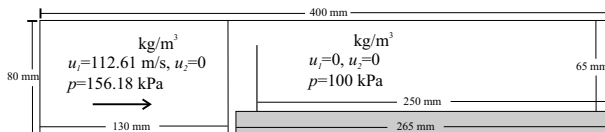
- ▶ SAMR base mesh $320 \times 64(\times 2)$, $r_{1,2} = 2$
- ▶ Intel 3.4GHz Xeon dual processors, GB Ethernet interconnect
 - ▶ Beam-FSI: 12.25 h CPU on 3 fluid CPU + 1 solid CPU
 - ▶ FEM-FSI: 322 h CPU on 14 fluid CPU + 2 solid CPU



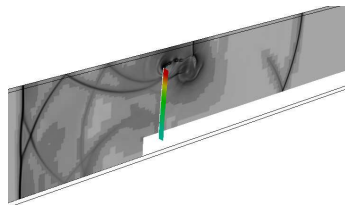
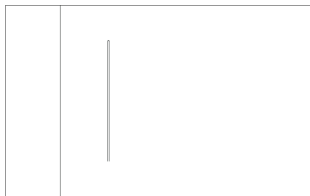
Shock-driven elastic panel motion

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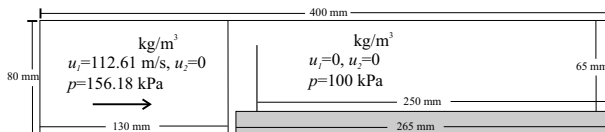


$t = 0.43$ ms after impact

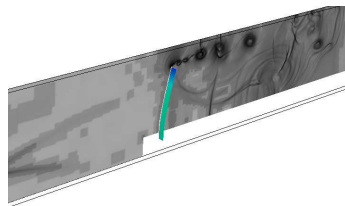
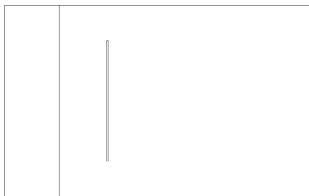
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$t = 1.56$ ms after impact

Detonation-driven plastic deformation

Chapman-Jouguet detonation in a tube filled with a stoichiometric ethylene and oxygen ($\text{C}_2\text{H}_4 + 3 \text{O}_2$, 295 K) mixture. Euler equations with single exothermic reaction $A \rightarrow B$

$$\partial_t \rho + \partial_{x_n}(\rho u_n) = 0, \quad \partial_t(\rho u_k) + \partial_{x_n}(\rho u_k u_n + \delta_{kn} p) = 0, \quad k = 1, \dots, d$$

$$\partial_t(\rho E) + \partial_{x_n}(u_n(\rho E + p)) = 0, \quad \partial_t(Y\rho) + \partial_{x_n}(Y\rho u_n) = \psi$$

with

$$p = (\gamma - 1)\left(\rho E - \frac{1}{2}\rho u_n u_n - \rho Y q_0\right) \quad \text{and} \quad \psi = -kY\rho \exp\left(\frac{-E_A\rho}{p}\right)$$

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modeled with heuristic detonation model by
[Mader, 1979]

$$V := \rho^{-1}, \quad V_0 := \rho_0^{-1}, \quad V_{\text{CJ}} := \rho_{\text{CJ}}$$

$$Y' := 1 - (V - V_0)/(V_{\text{CJ}} - V_0)$$

If $0 \leq Y' \leq 1$ and $Y > 10^{-8}$ then

If $Y < Y'$ and $Y' < 0.9$ then $Y' := 0$

If $Y' < 0.99$ then $p' := (1 - Y')p_{\text{CJ}}$

else $p' := p$

$$\rho_A := Y'\rho$$

$$E := p'/(\rho(\gamma - 1)) + Y'q_0 + \frac{1}{2}u_n u_n$$

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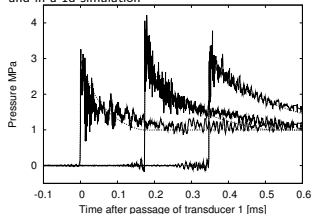
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Comparison of the pressure traces in the experiment and in a 1d simulation



Tube with flaps

► Fluid: VanLeer FVS

- Detonation model with $\gamma = 1.24$, $p_{CJ} = 3.3$ MPa, $D_{CJ} = 2376$ m/s
- AMR base level: $104 \times 80 \times 242$, $r_{1,2} = 2$, $r_3 = 4$
- $\sim 4 \cdot 10^7$ cells instead of $7.9 \cdot 10^9$ cells (uniform)
- Tube and detonation fully refined
- Thickening of 2D mesh: 0.81 mm on both sides (real 0.445 mm)

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► Solid: thin-shell solver by F. Cirak

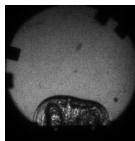
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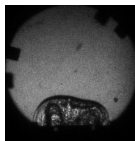


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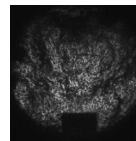
0.032 ms



0.030 ms

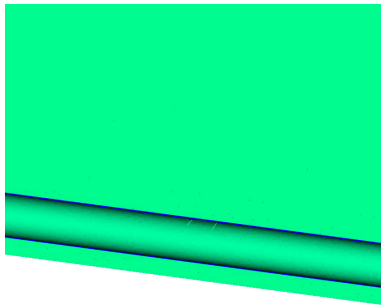


0.212 ms



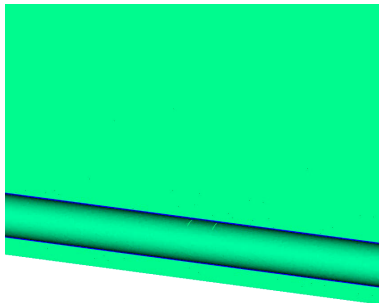
0.210 ms

Tube with flaps: results

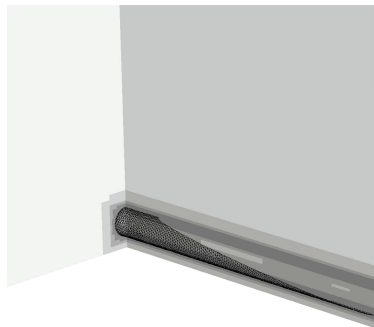


Fluid density and displacement in y-direction in solid

Tube with flaps: results



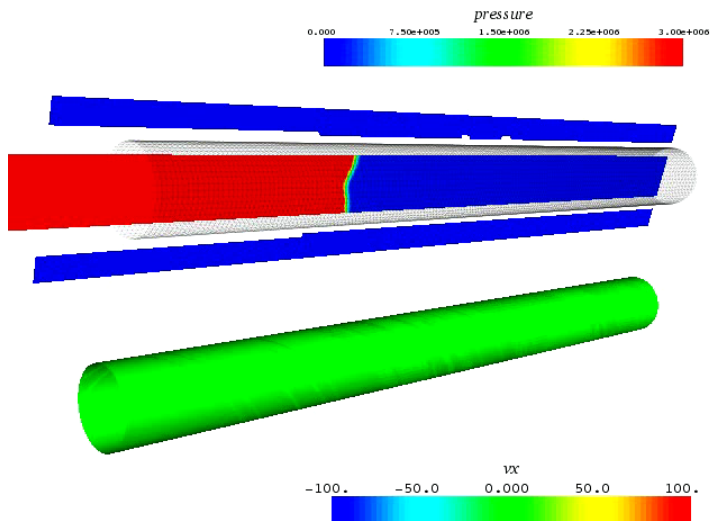
Fluid density and displacement in y-direction in solid



Schlieren plot of fluid density on refinement levels

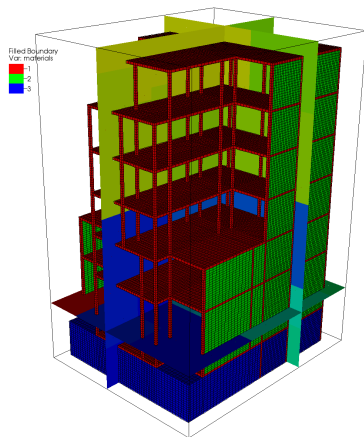
[Cirak et al., 2007]

Coupled fracture simulation



Blast explosion in a multistory building

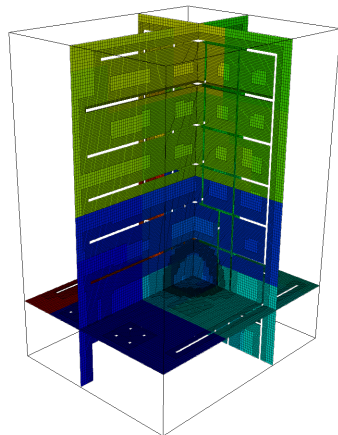
- ▶ 20 m × 40 m × 25 m seven-story building similar to [Luccioni et al., 2004]
- ▶ Spherical energy deposition \equiv 400 kg TNT, $r = 0.5$ m in lobby of building
- ▶ SAMR: $80 \times 120 \times 90$ base level, three additional levels $r_{1,2} = 2$, $l_{\text{fsi}} = 1$, $k = 1$
- ▶ Simulation with ground: 1,070 coupled time steps, 830 h CPU (~ 25.9 h wall time) on 31+1 cores
- ▶ $\sim 8,000,000$ cells instead of 55,296,000 (uniform)
- ▶ 69,709 hexahedral elements and with material parameters. [Deiterding and Wood, 2013]



	ρ_s [kg/m ³]	σ_0 [MPa]	E_T [GPa]	β	K [GPa]	G [GPa]	$\bar{\epsilon}^P$	p_f [MPa]
Columns	2010	50	11.2	1.0	21.72	4.67	0.02	-30
Walls	2010	25	11.2	1.0	6.22	4.67	0.01	-15

Blast explosion in a multistory building

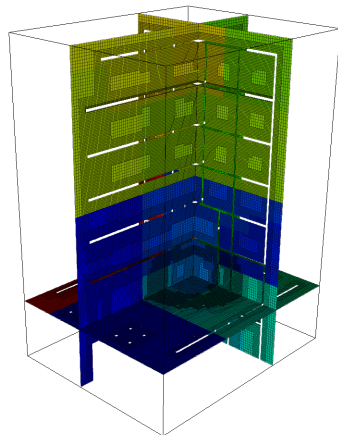
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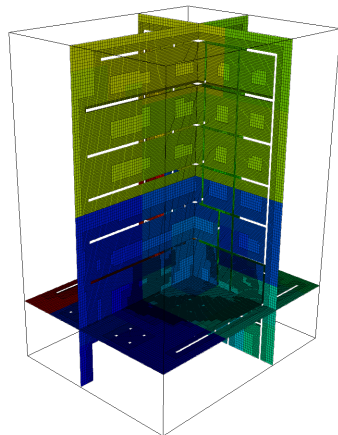
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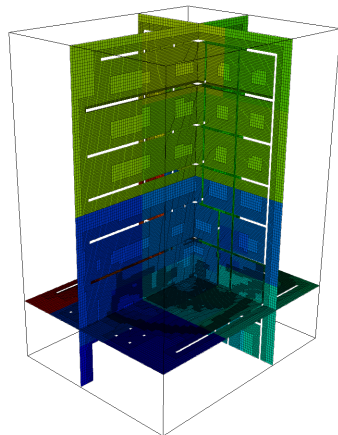
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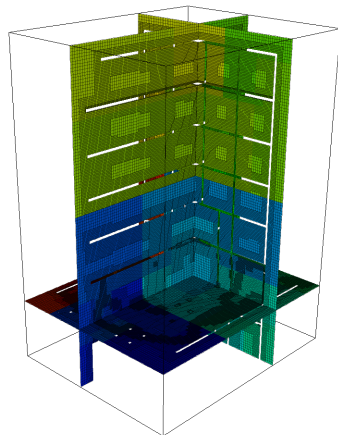
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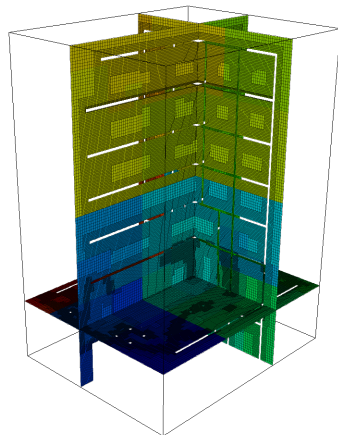
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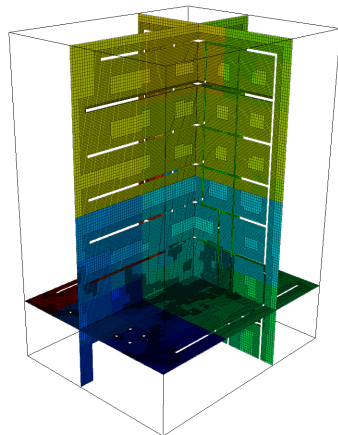
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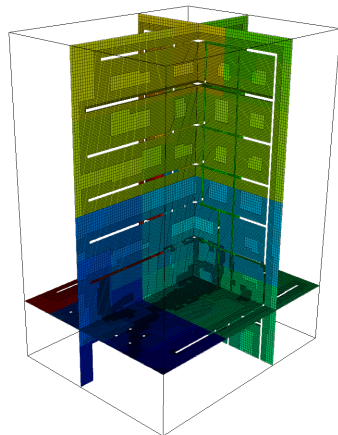
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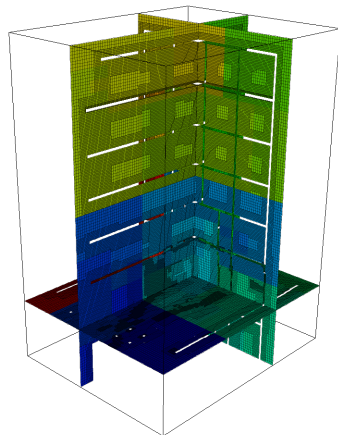
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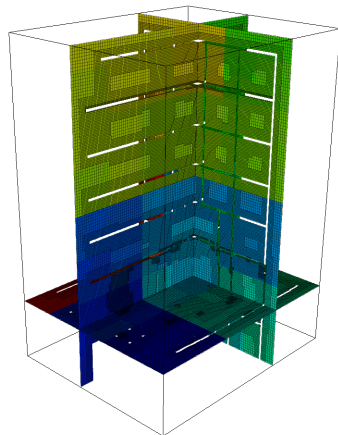
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Walls	2010	25	11.2	1.0	6.22	4.67	0.01	-15

Blast explosion in a multistory building

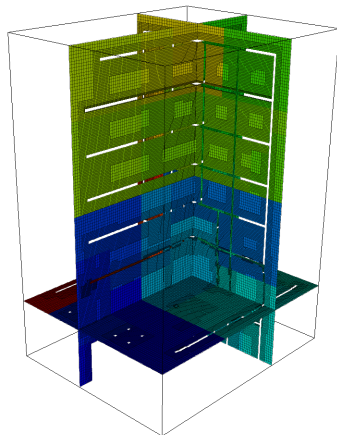
- ▶ 20 m × 40 m × 25 m seven-story building similar to [Luccioni et al., 2004]
- ▶ Spherical energy deposition \equiv 400 kg TNT, $r = 0.5$ m in lobby of building
- ▶ SAMR: $80 \times 120 \times 90$ base level, three additional levels $r_{1,2} = 2$, $l_{\text{fsi}} = 1$, $k = 1$
- ▶ Simulation with ground: 1,070 coupled time steps, 830 h CPU (~ 25.9 h wall time) on 31+1 cores
- ▶ $\sim 8,000,000$ cells instead of 55,296,000 (uniform)
- ▶ 69,709 hexahedral elements and with material parameters. [Deiterding and Wood, 2013]



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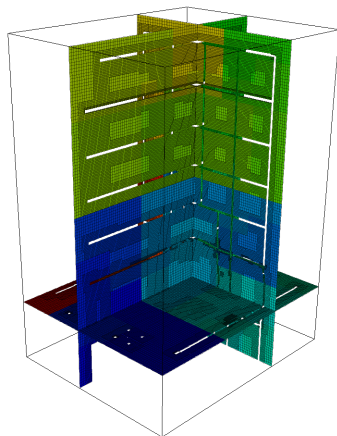
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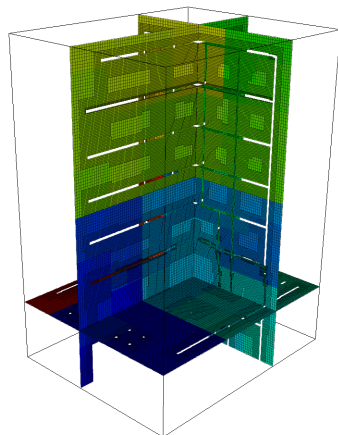
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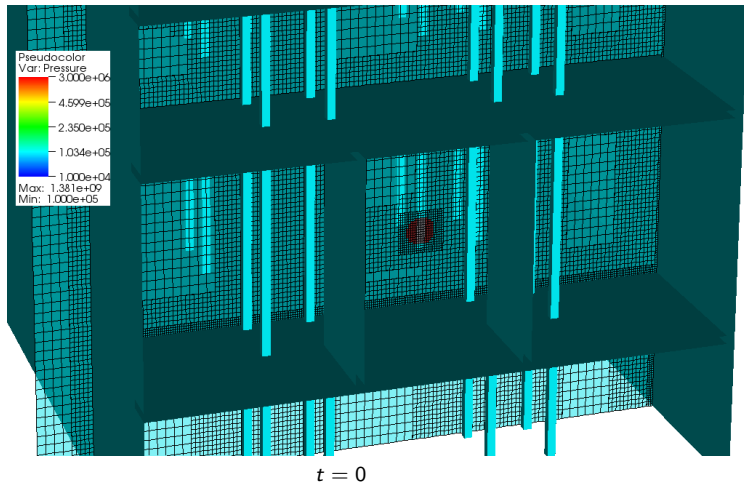
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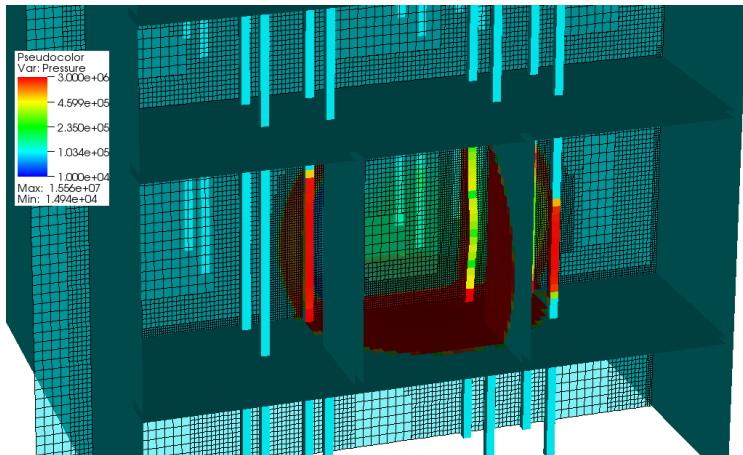


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Columns	2010	50	11.2	1.0	21.72	4.67	0.02	-30
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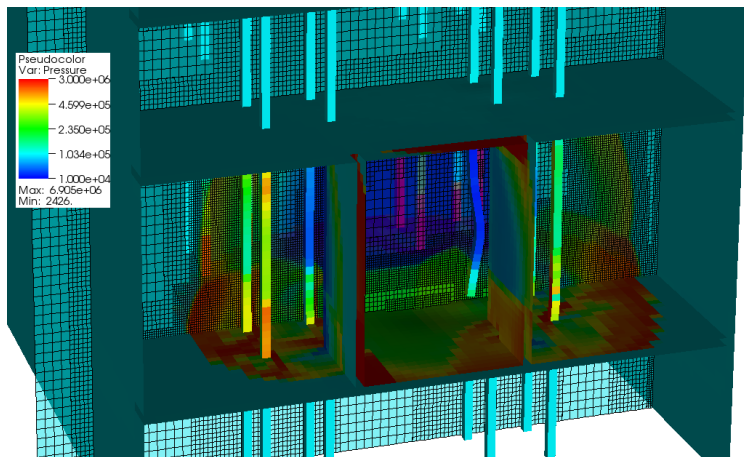
Blast explosion in a multistory building – II



Blast explosion in a multistory building – II

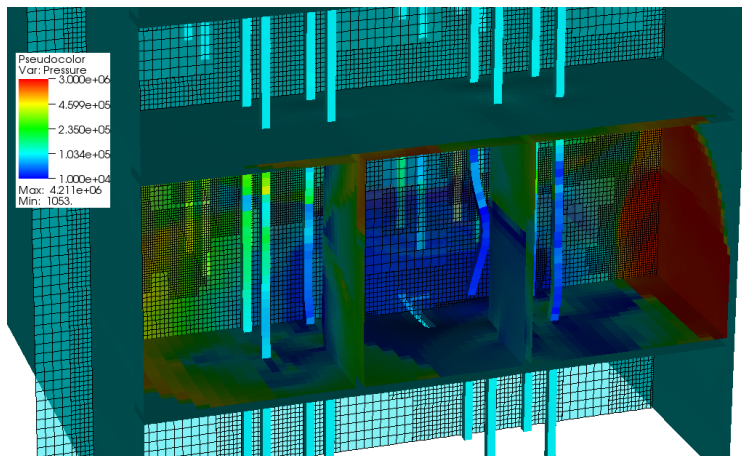


Blast explosion in a multistory building – II

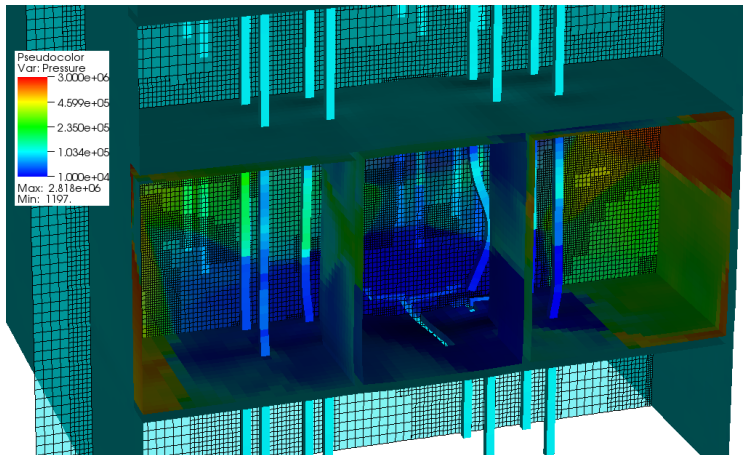


$t = 6.1 \text{ ms}$

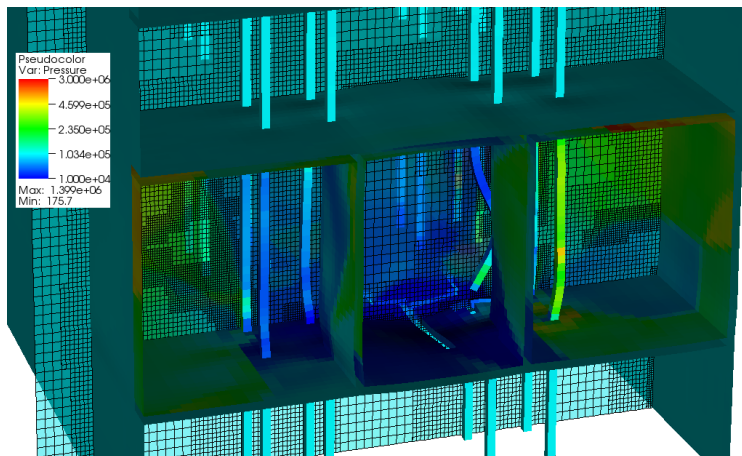
Blast explosion in a multistory building – II



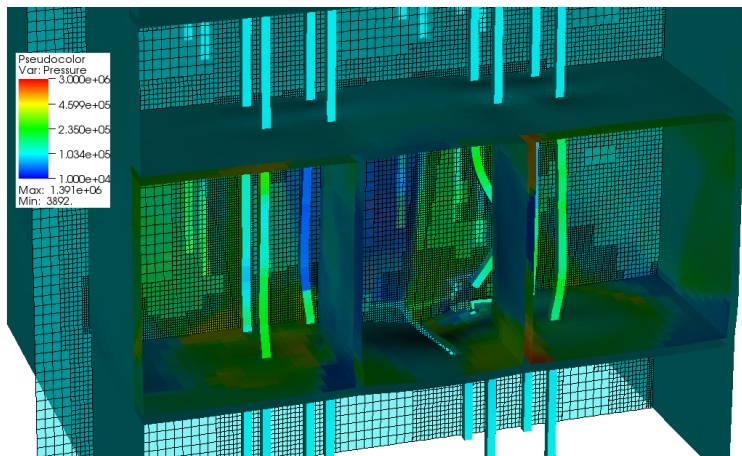
Blast explosion in a multistory building – II



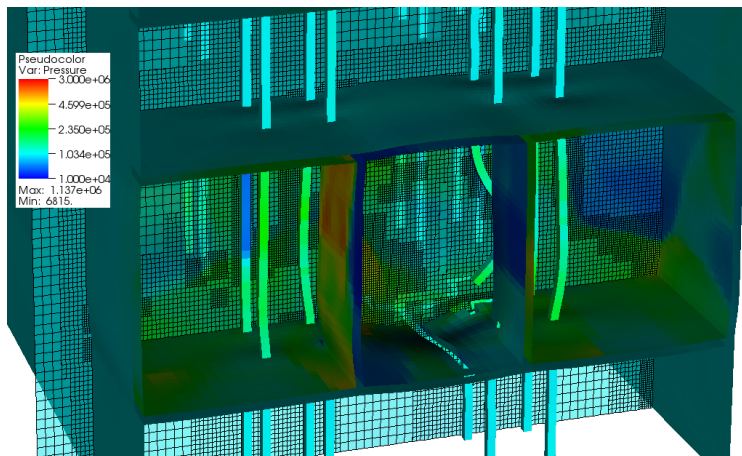
Blast explosion in a multistory building – II



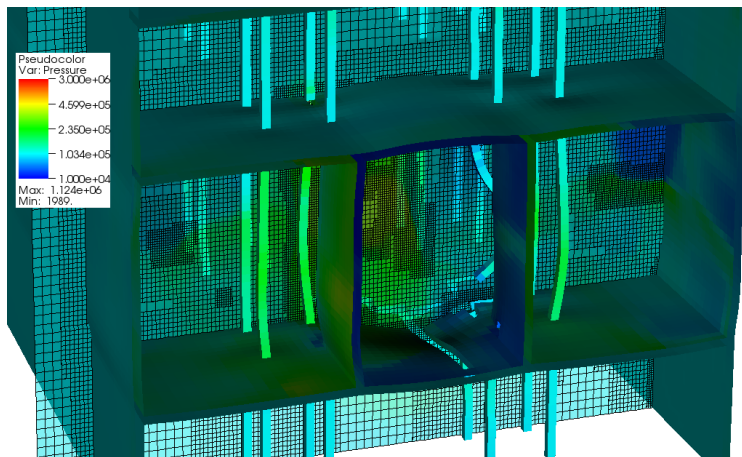
Blast explosion in a multistory building – II



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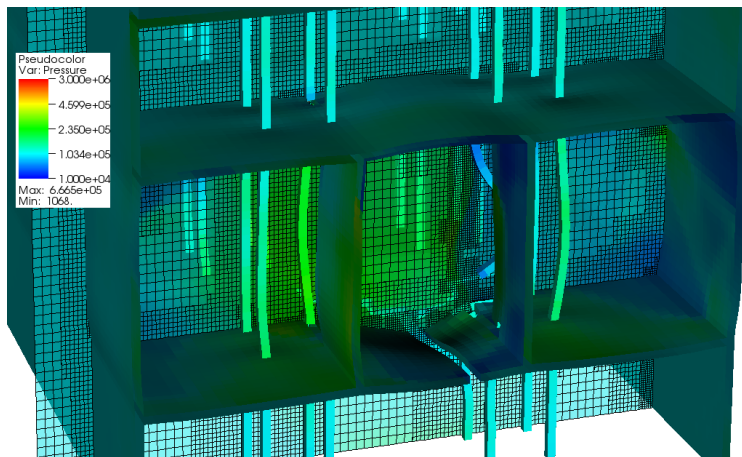


Blast explosion in a multistory building – II

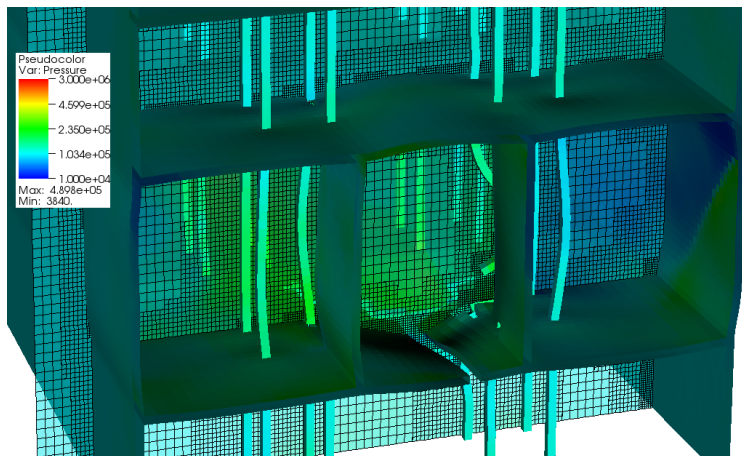


$t = 29.2 \text{ ms}$

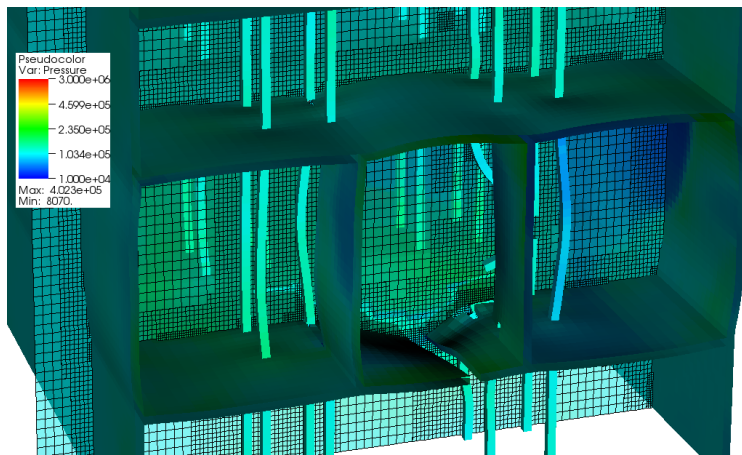
Blast explosion in a multistory building – II



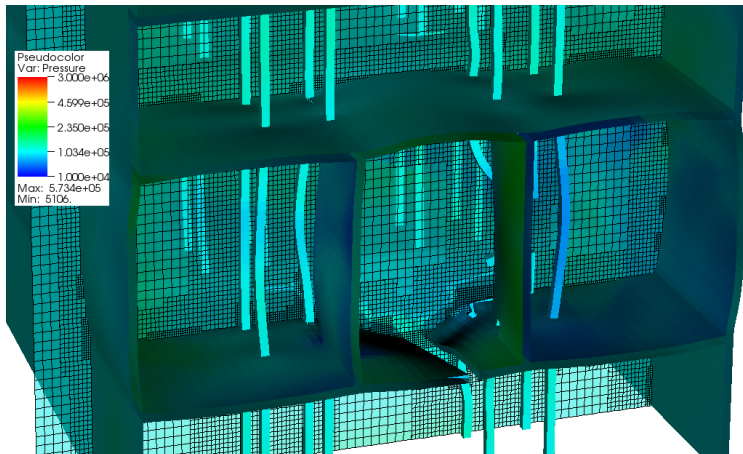
Blast explosion in a multistory building – II



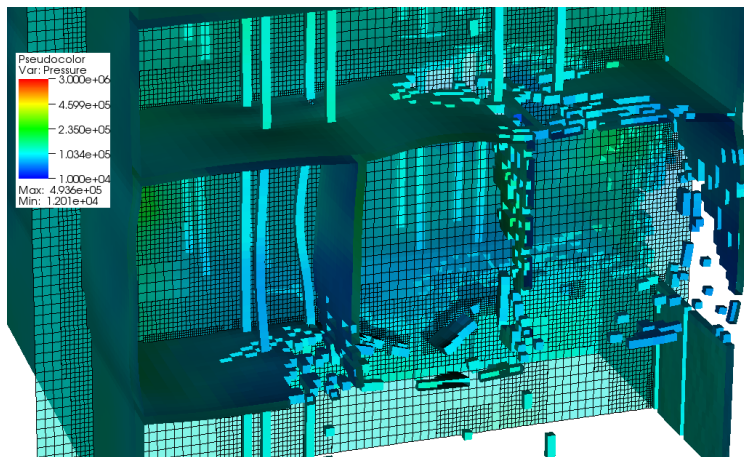
Blast explosion in a multistory building – II



Blast explosion in a multistory building – II

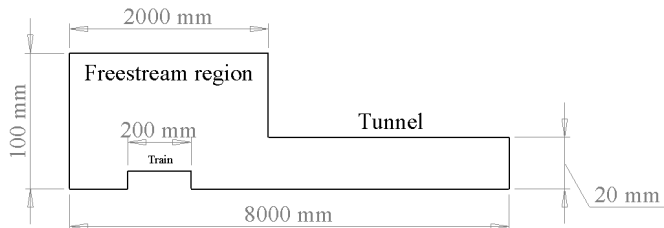


Blast explosion in a multistory building – II

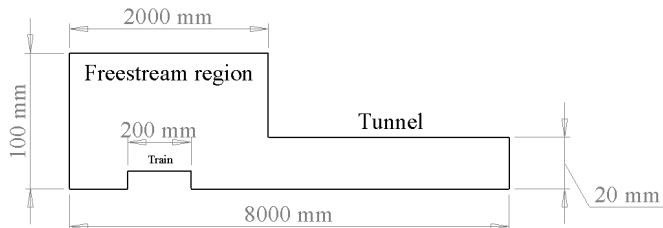


$t = 48.7$ ms

Laboratory tunnel simulator [Zonglin et al., 2002]



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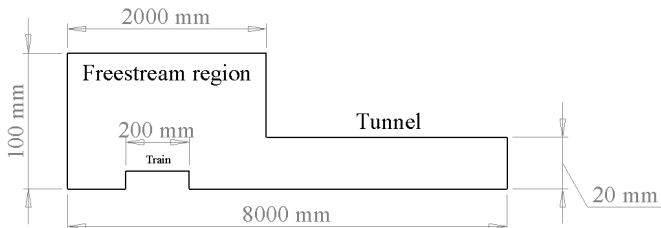


Model solves the inviscid Euler equations

$$\begin{aligned}\partial_t \rho + \nabla \cdot (\rho \mathbf{u}) &= 0 \\ \partial_t (\rho \mathbf{u}) + \nabla \cdot (\rho \mathbf{u} \otimes \mathbf{u}) + \nabla p &= 0 \\ \partial_t (\rho E) + \nabla \cdot ((\rho E + p) \mathbf{u}) &= 0\end{aligned}$$

with $p = (\gamma - 1)(\rho E - \frac{1}{2} \rho \mathbf{u}^T \mathbf{u})$

Laboratory tunnel simulator [Zonglin et al., 2002]



Model solves the inviscid Euler equations

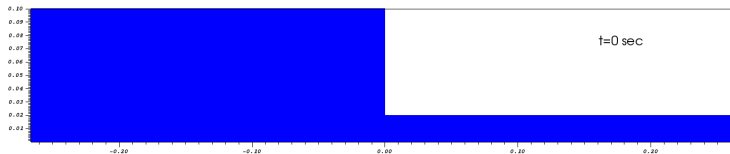
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with $p = (\gamma - 1)(\rho E - \frac{1}{2} \rho \mathbf{u}^T \mathbf{u})$

- ▶ Two-dimensional axi-symmetric computation
- ▶ $p_0 = 100 \text{ kPa}$, $\rho_0 = 1.225 \text{ kg/m}^3$, $\gamma = 1.4$
- ▶ Roe shock-capturing scheme blended with HLL
- ▶ 2nd order accuracy achieved with MUSCL-Hancock method

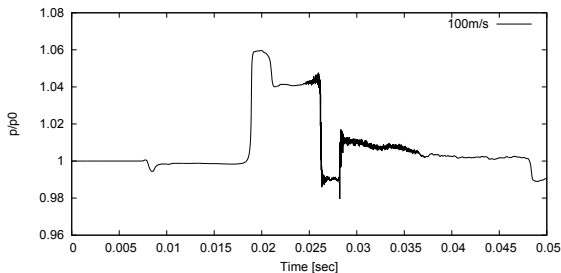
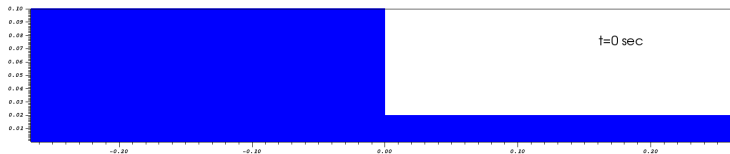
Basic phenomena – $v_0 = 100$ m/s

- ▶ 800×25 mesh with Cartesian cut-out (200, 5) to (800, 25)
- ▶ 2 level of additional refinement by factor 2



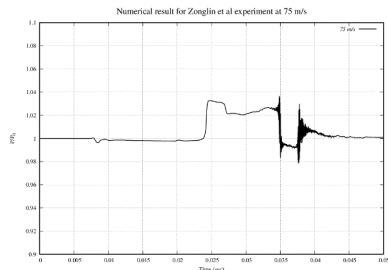
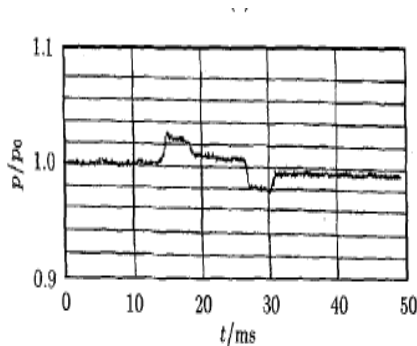
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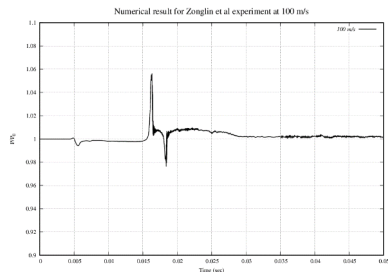
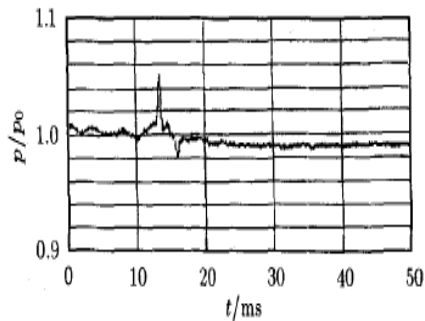
Pressure record at location (1020 mm, 20 mm) inside tunnel

Comparison with experiment – I



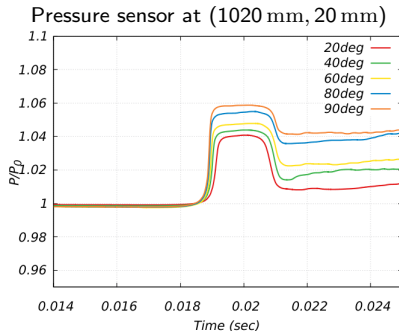
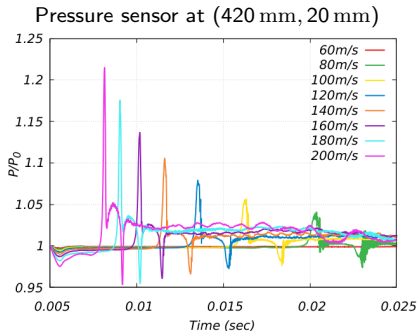
Pressure record at (1020 mm, 20 mm) for $v_0 = 75$ m/s. Experiment (left) and AMROC (right)

Comparison with experiment – I



Pressure record at (40 mm, 20 mm) for model velocity $v_0 = 100$ m/s. Experiment (left) and AMROC (right)

Variation of velocity and nose half angle



- ▶ Dependence on v_0^2 is the dynamic pressure influence (left)
- ▶ For constant blockage ratio and body velocity, using more pointed noses alleviates the maximal pressure level (right, nose half angle varied)
- ▶ For $v_0 \approx 140$ m/s a shock wave (tunnel boom) can be observed. Sharper noses also delay this phenomenon.

NGT2 prototype setup

- ▶ Next Generation Train 2 (NGT2) geometry by the German Aerospace Centre (DLR) [Fragner and Deiterding, 2016, Fragner and Deiterding, 2017]
- ▶ Mirrored train head of length ~ 60 m, no wheels or tracks, train models 0.17 m above ground above the ground level.
- ▶ Train velocities 100 m/s and -100 m/s, middle axis 6 m apart, initial distance between centers 200 m
- ▶ Base mesh of $360 \times 40 \times 30$ for domain of $360 \text{ m} \times 40 \text{ m} \times 30 \text{ m}$
- ▶ Two/three additional levels, refined by $r_{1,2,3} = 2$. Refinement based on pressure gradient and level set and regenerated at every coarse time step. Parallel redistribution at every level-0 time step.
- ▶ On 96 cores Intel Xeon E5-2670 2.6 GHz a final $t_e = 3 \text{ sec}$ was reached after 12,385 sec / 43,395 sec wall time, i.e., 330 h and 1157 h CPU

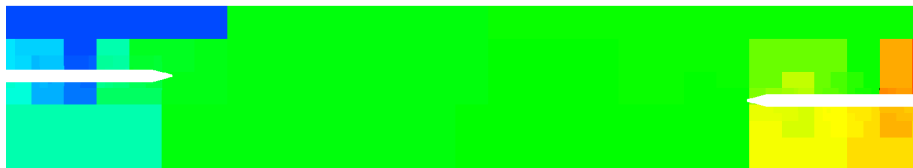


Passing in open space – AMR and dynamic distribution

Domains of three-level refinement



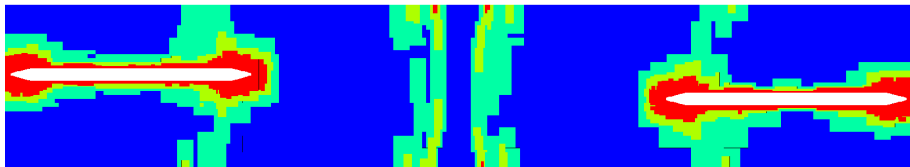
Distribution to 96 processors



Enlargement of domain center shown

Passing in open space – AMR and dynamic distribution

Domains of three-level refinement



Distribution to 96 processors



Enlargement of domain center shown

Passing in open space – AMR and dynamic distribution

Domains of three-level refinement



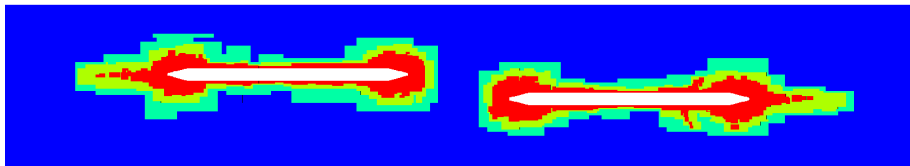
Distribution to 96 processors



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Passing in open space – AMR and dynamic distribution

Domains of three-level refinement



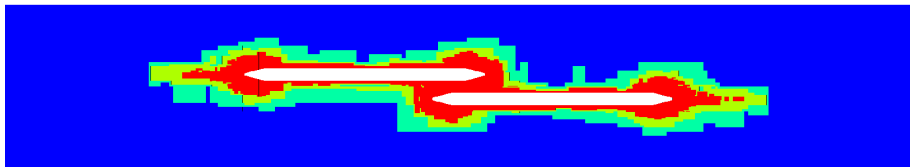
Distribution to 96 processors



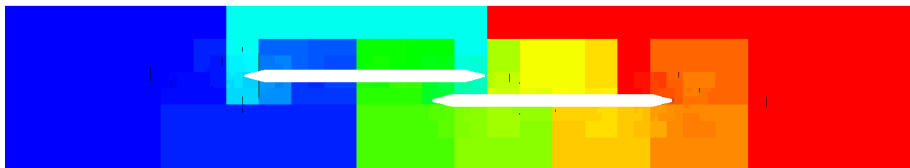
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Passing in open space – AMR and dynamic distribution

Domains of three-level refinement



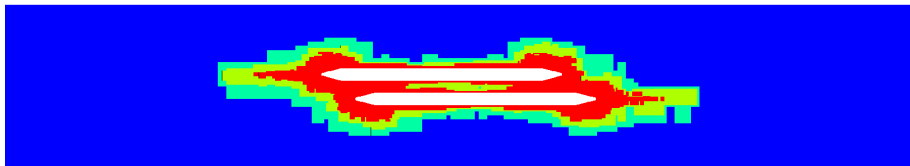
Distribution to 96 processors



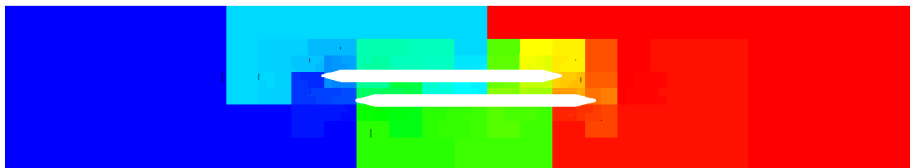
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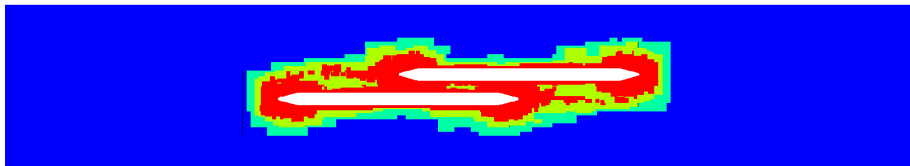
Distribution to 96 processors



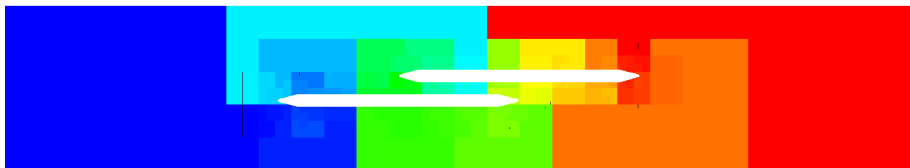
Enlargement of domain center shown

Passing in open space – AMR and dynamic distribution

Domains of three-level refinement



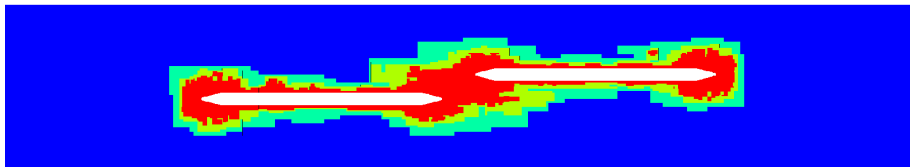
Distribution to 96 processors



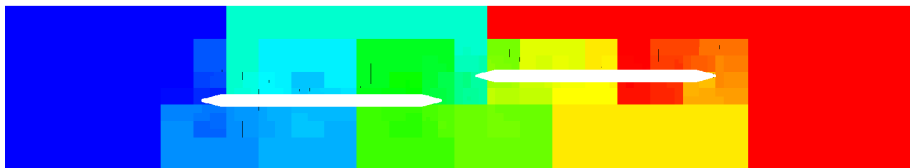
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Passing in open space – AMR and dynamic distribution

Domains of three-level refinement



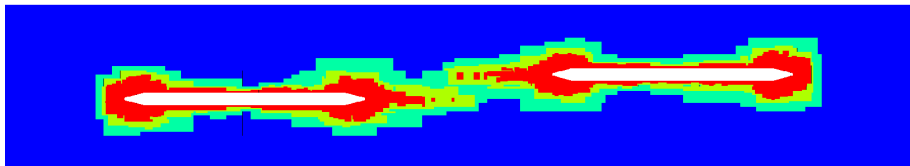
Distribution to 96 processors



Enlargement of domain center shown

Passing in open space – AMR and dynamic distribution

Domains of three-level refinement

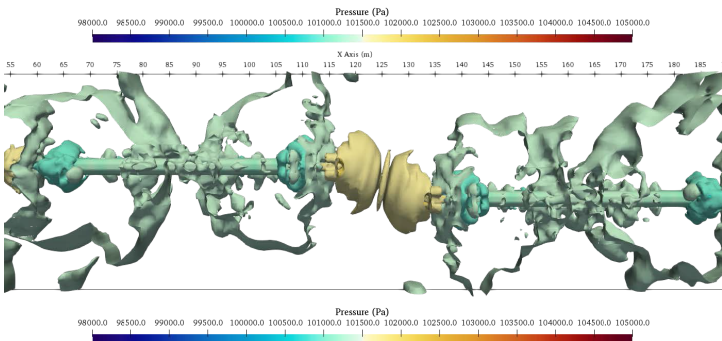
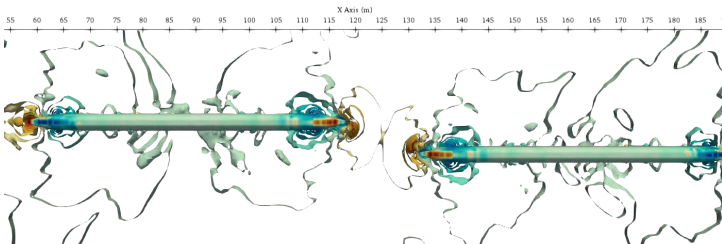


Distribution to 96 processors

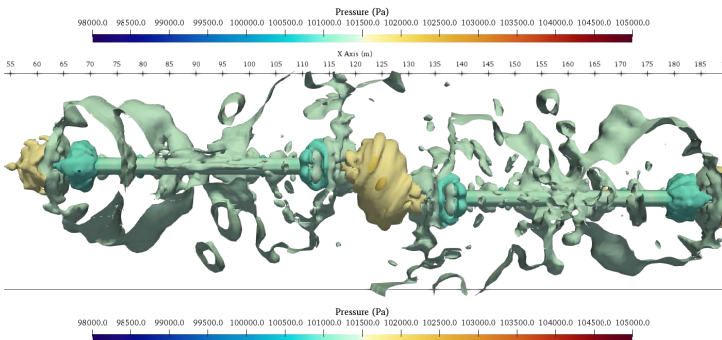
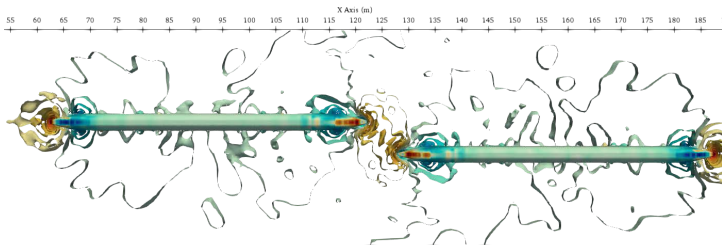


Enlargement of domain center shown

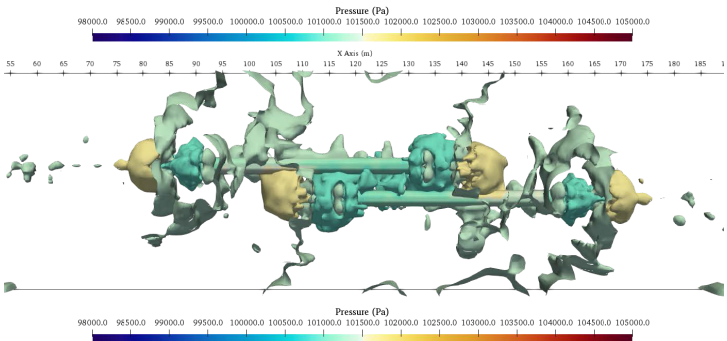
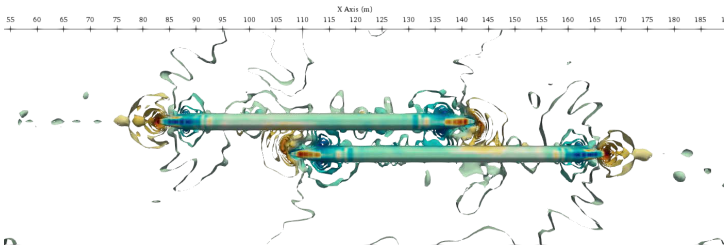
Pressure isosurfaces



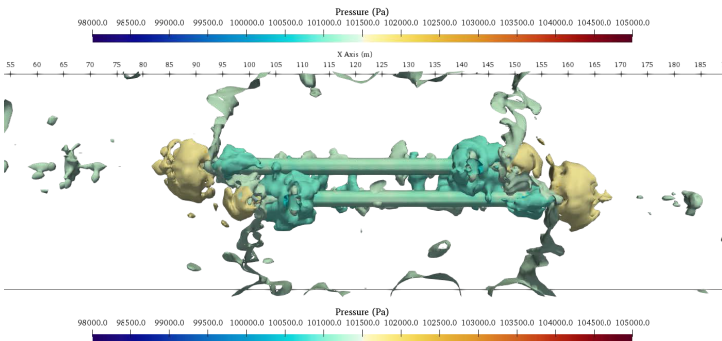
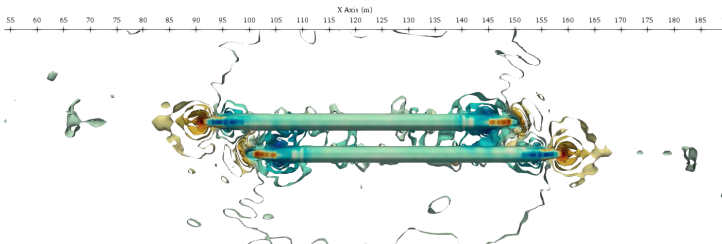
Pressure isosurfaces



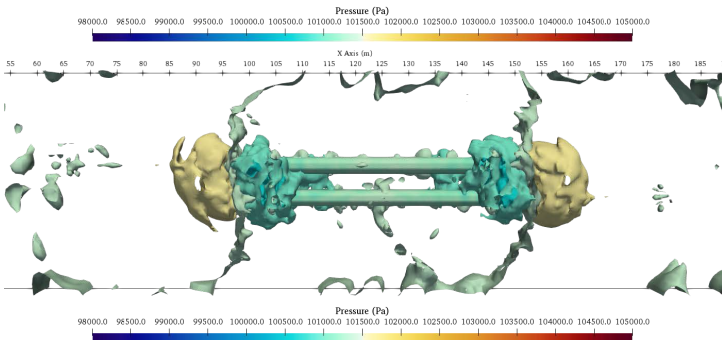
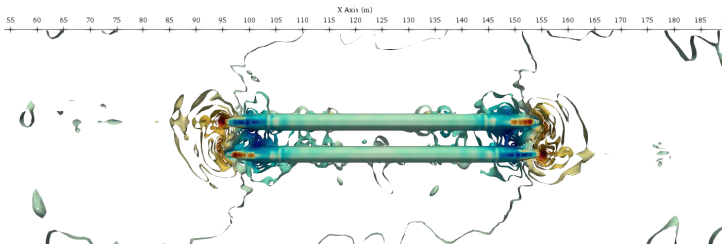
Pressure isosurfaces



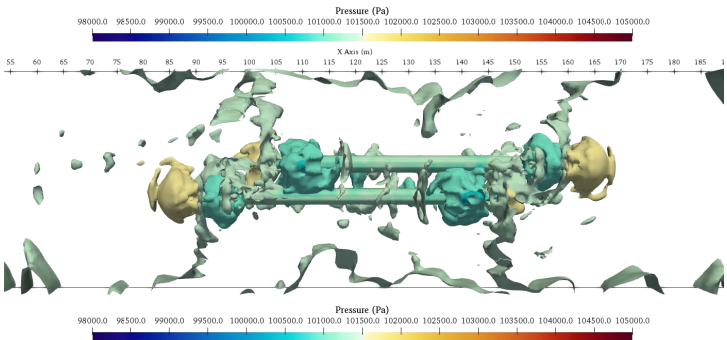
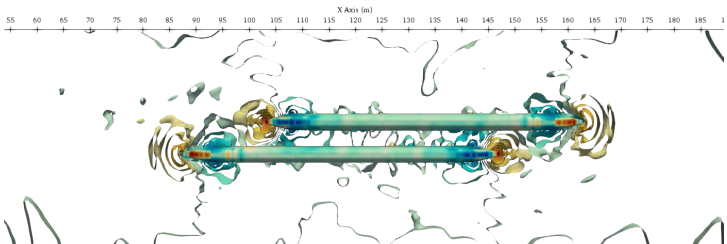
Pressure isosurfaces



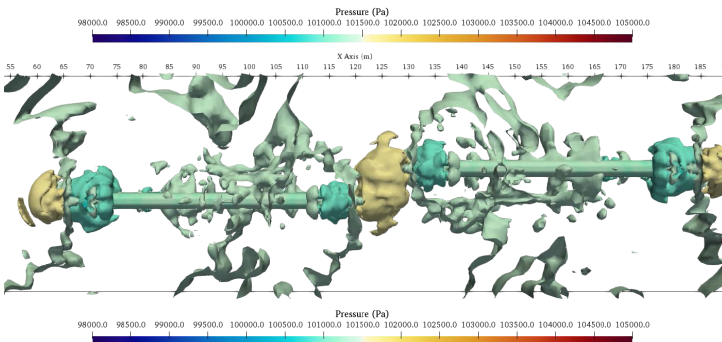
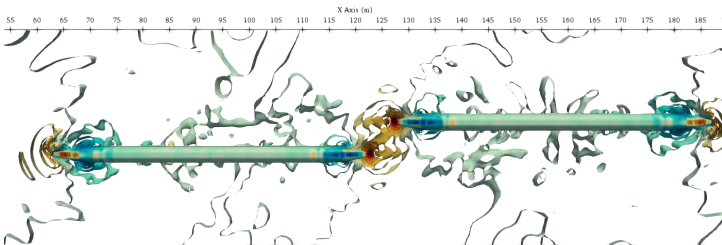
Pressure isosurfaces



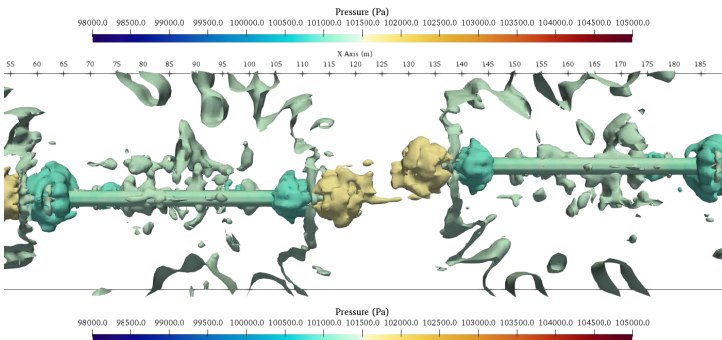
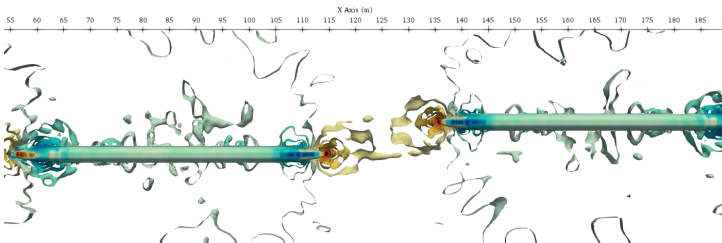
Pressure isosurfaces



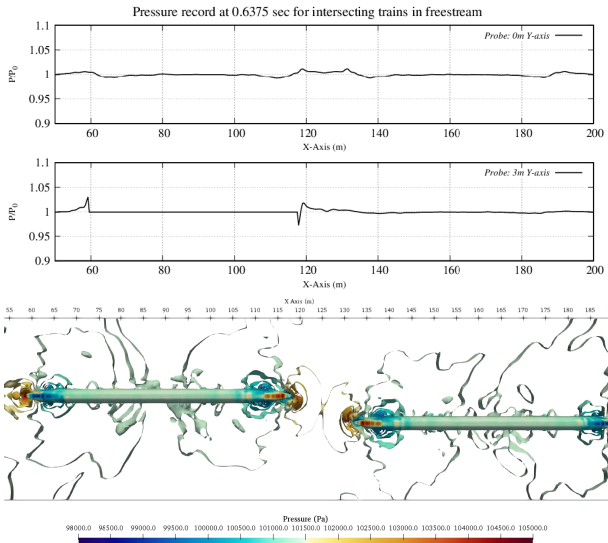
Pressure isosurfaces



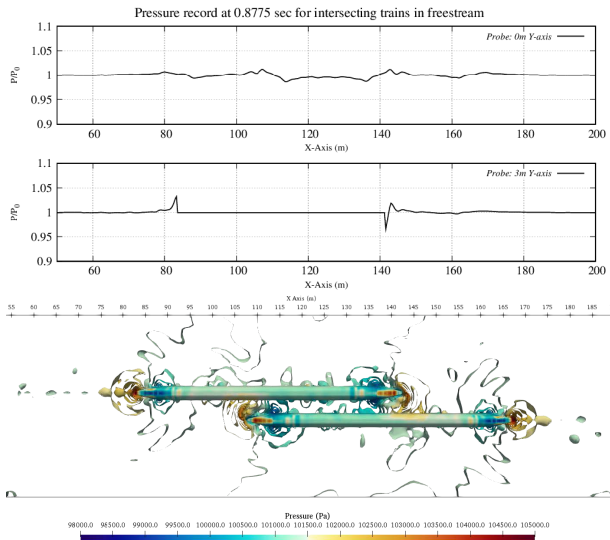
Pressure isosurfaces



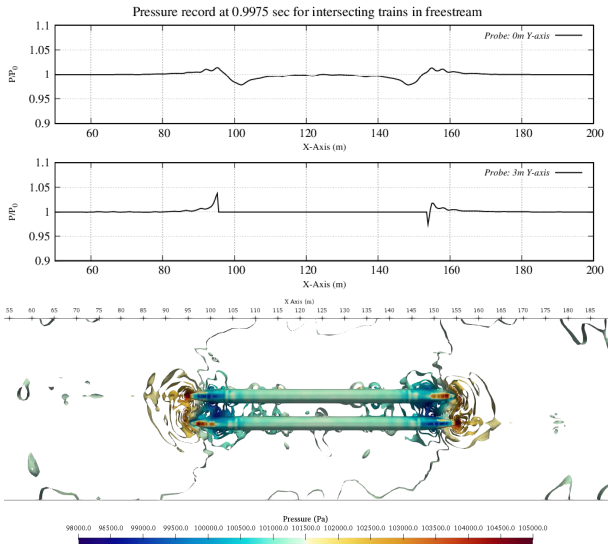
Pressure transects



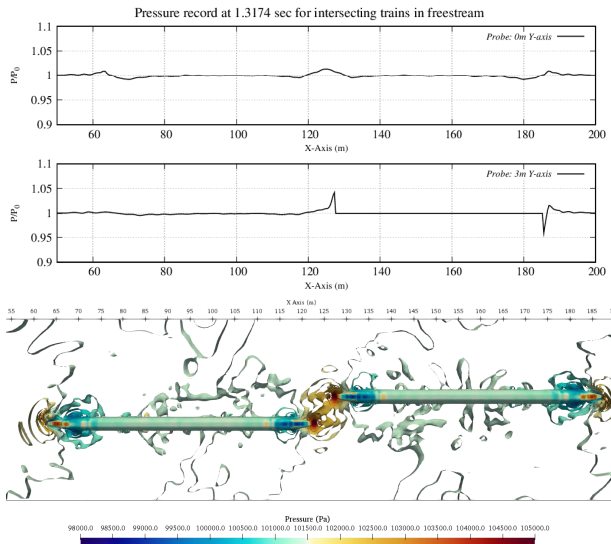
Pressure transects



Pressure transects

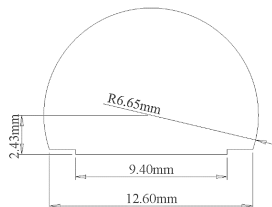


Pressure transects



Setup with realistic tunnel shape

- ▶ Two NGT2 trains again at velocities 100 m/s and -100 m/s
- ▶ Prototype straight double track tunnel of 640 m length, initial distance between centers of trains 820 m
- ▶ Base mesh of $1060 \times 36 \times 24$ for domain of $1060 \text{ m} \times 36 \text{ m} \times 24 \text{ m}$, three levels refined by $r_{1,2,3} = 2$
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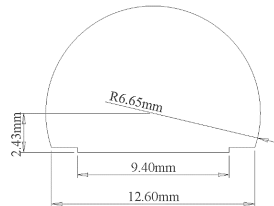


Tunnel shape



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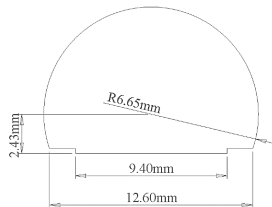


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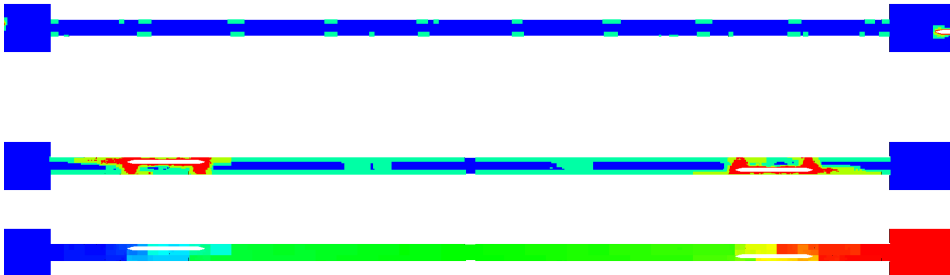


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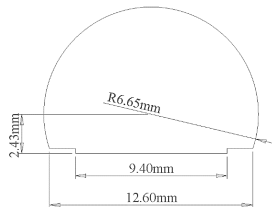


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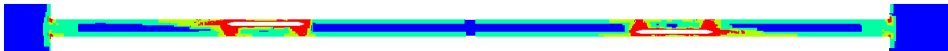


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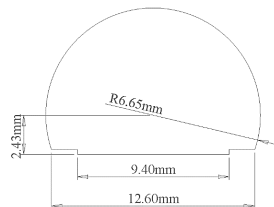


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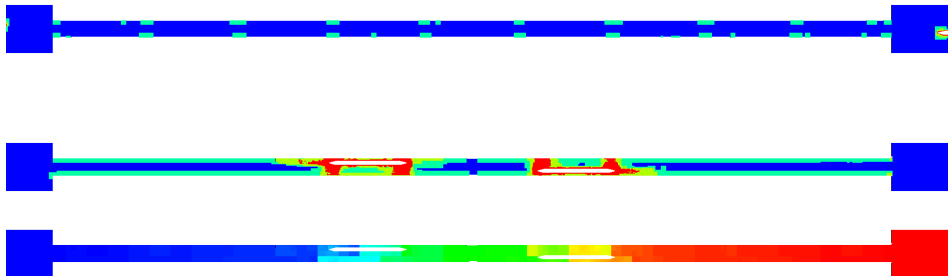


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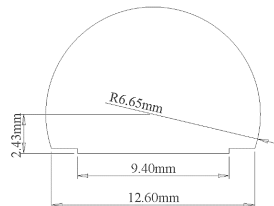


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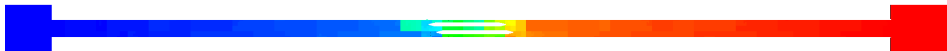


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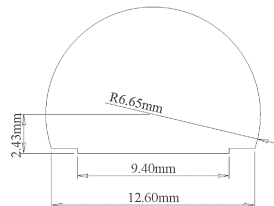


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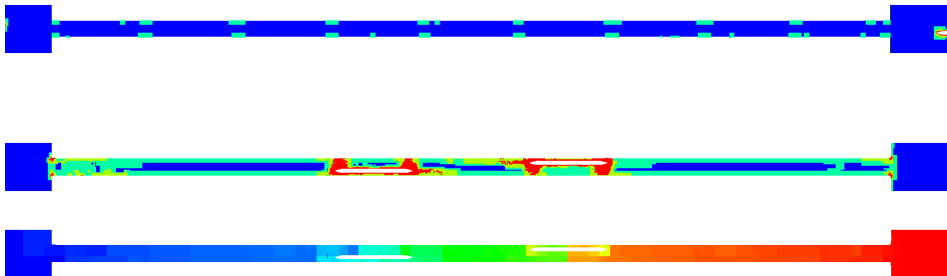


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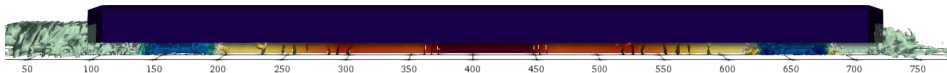
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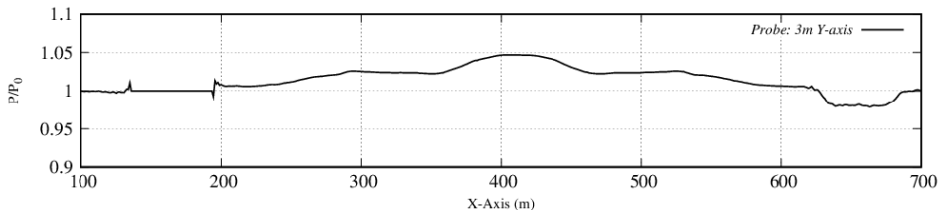
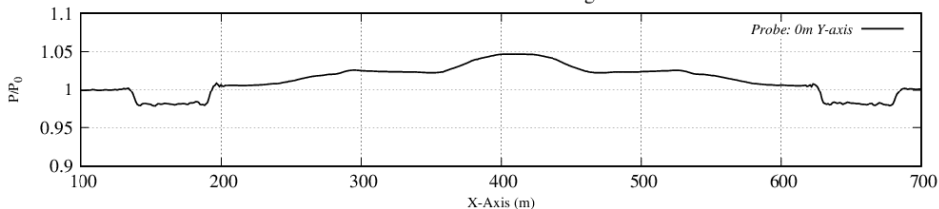
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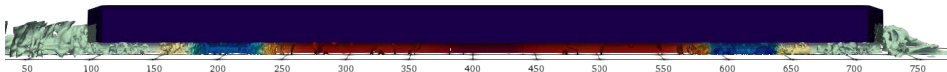
Pressure transects



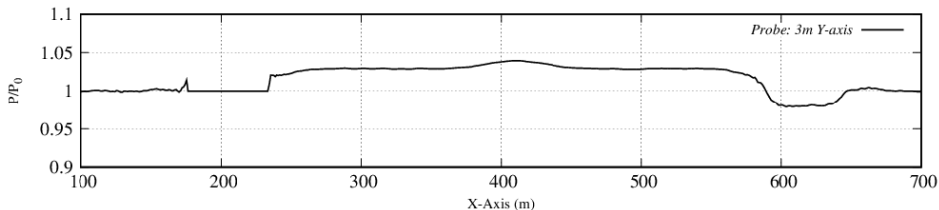
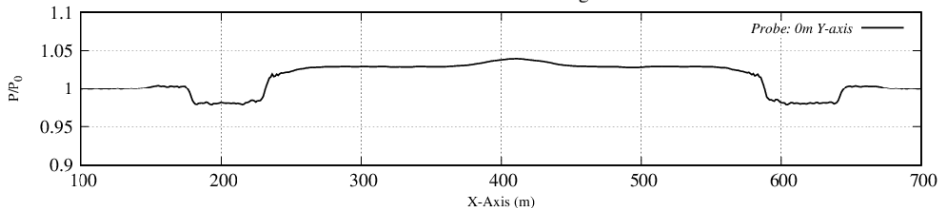
Pressure record at 1.39875 seconds for trains intersecting inside a double-track tunnel



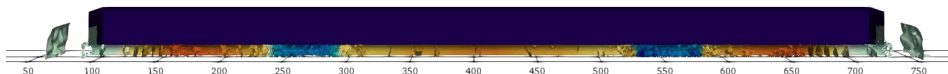
Pressure transects



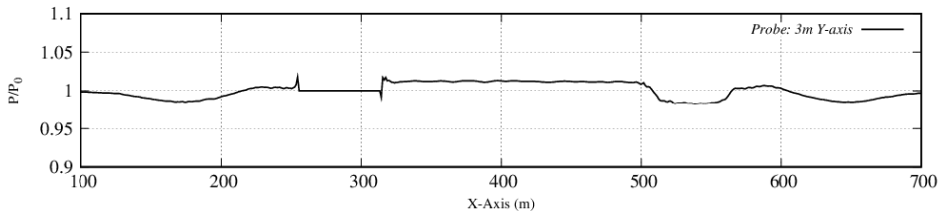
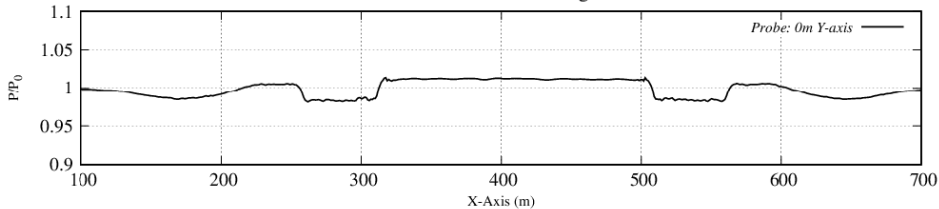
Pressure record at 1.7987 seconds for trains intersecting inside a double-track tunnel



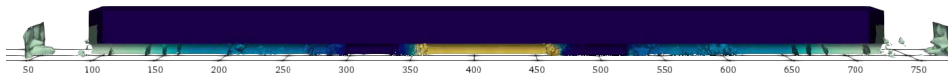
Pressure transects



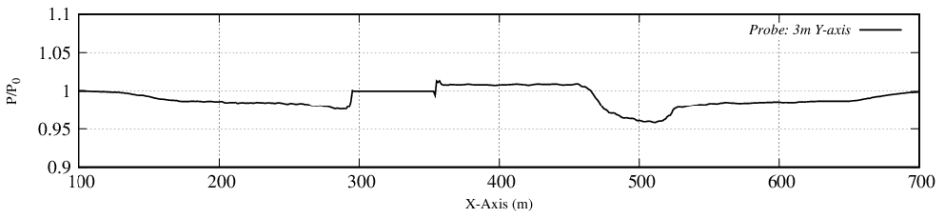
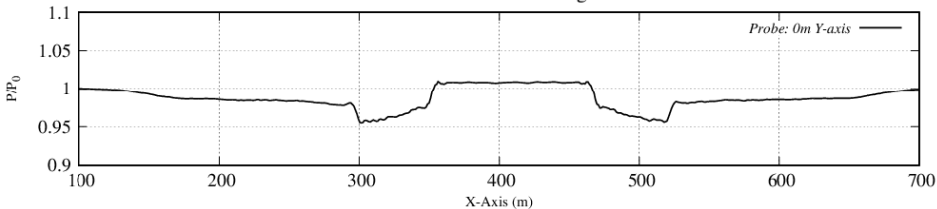
Pressure record at 2.5987 seconds for trains intersecting inside a double-track tunnel



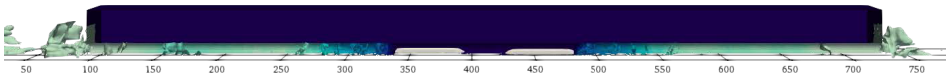
Pressure transects



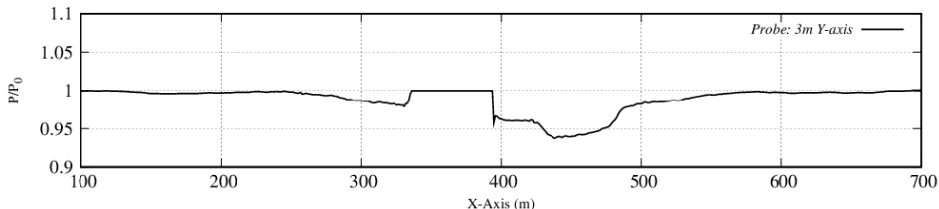
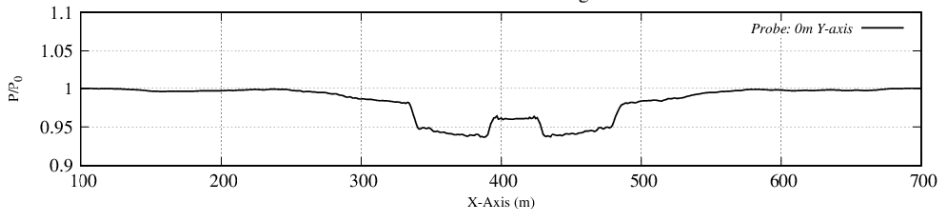
Pressure record at 2.9988 seconds for trains intersecting inside a double-track tunnel



Pressure transects



Pressure record at 3.3987 seconds for trains intersecting inside a double-track tunnel



Conclusions – compressible flow aerodynamics

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- ▶ Multi-resolution and fluid-structure coupling problems can be tackled without expensive mesh regeneration
 - ▶ Level set approach easily handles large motions, element failure and removal
 - ▶ Dynamic adaptation ensures high resolution at embedded boundaries and essential flow features
- ▶ Aerodynamics of bodies with large motion are easily accessible
 - ▶ Current inviscid approach predicts maximal overpressure in front of trains reliably
 - ▶ For predicting the flow around entire trains, the boundary layer growing over the train body needs to be considered.
 - ▶ AMROC solvers for the compressible Navier-Stokes equations and even LES are already available, however, for this particular application a turbulent wall function on the embedded boundary first needs to be implemented. Such a wall function is currently work-in-progress for the LBM-LES solver.

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